

1 Log-normal Kalman filter for assimilating 2 phase-space density data in the radiation belts

D. Kondrashov,¹ M. Ghil^{1,3} and Y. Shprits,^{1,3}

D. Kondrashov, Department of Atmospheric and Oceanic Sciences, 405 Hilgard Ave, Box 951565, 7127 Math Sciences Bldg. UCLA Los Angeles, CA 90095-1565, U.S.A. (dkon-
dras@atmos.ucla.edu)

¹Department of Atmospheric and Oceanic Sciences and Institute of Geophysics and Planetary Physics, University of California, Los Angeles, U.S.A.

²Geosciences Department and Laboratoire de Météorologie Dynamique (CNRS and IPSL), Ecole Normale Supérieure, F-75231 Paris Cedex 05, FRANCE.

³Department of Earth and Space Sciences and Institute of Geophysics and Planetary Physics, University of California, Los Angeles, U.S.A.

3 **Abstract.** Data assimilation combines a physical model with sparse ob-
4 servations, and has become an increasingly important tool for scientists and
5 engineers in the design, operation and use of satellites and other high-technology
6 systems in near-Earth's space environment. Of particular importance is pre-
7 dicting fluxes of high-energy particles in the Van Allen radiation belts, since
8 these fluxes can damage space-borne platforms and instruments during strong
9 geomagnetic storms. In transiting from a research setting to operational pre-
10 diction of these fluxes, improved data assimilation is of the essence. The present
11 study is motivated by the fact that phase-space densities (PSDs) of high-energy
12 electrons in the outer radiation belt — both simulated and observed — are
13 subject to spatio-temporal variations that span several orders of magnitude.
14 Standard data assimilation methods that are based on least-squares mini-
15 mization of normally distributed errors may not be adequate for handling
16 the range of these variations. We propose herein a modification of Kalman
17 filtering that uses a log-transformed, one-dimensional radial diffusion model
18 for the PSDs and includes parameterized losses. The proposed methodology
19 is first verified on model-simulated, synthetic data and then applied to ac-
20 tual satellite measurements. When the model errors are sufficiently smaller
21 than observational errors, our methodology can significantly improve anal-
22 ysis and prediction skill for the PSDs compared to those of the standard Kalman
23 filter formulation. This improvement is documented by monitoring the vari-
24 ance of the innovation sequence.

1. Introduction and Motivation

1.1. Data assimilation and operational prediction

25 In the process of moving from research to operations in the study of the Van Allen
26 radiation belts, it is of the essence to properly understand and further improve data as-
27 similation methodology — as applied to the filtering, smoothing and prediction of electron
28 fluxes and phase space density (PSD) fields. The modern uses of data assimilation in the
29 geosciences go back to the introduction of meteorological satellites and their application
30 to numerical weather prediction (NWP) in the late 1960s and the 1970s (*Charney et al.*
31 [1969]). It was *Bjerknes* [1904], following the earlier ideas of H. Helmholtz and others,
32 who formulated the NWP problem as an initial-value problem for the partial differential
33 equations that govern large-scale atmospheric flow.

34 As soon as the research group around J. von Neumann at the Institute for Advanced
35 Studies in Princeton started working on experimental NWP (*Charney et al.* [1950]), it
36 became apparent that the initial state of the atmosphere at any given time was known
37 only very partially and inaccurately. The World Weather Watch introduced by the World
38 Meteorological Organization after World War II was designed to provide as good a state
39 of the atmosphere as possible twice a day, at noon and midnight Greenwich Mean Time,
40 the so-called synoptic times. These synoptic (i.e., simultaneous) observations, however,
41 were too costly or impractical to provide adequate coverage of the weather over the entire
42 globe. The advent of asynchronous observations, via satellites and other unconventional
43 observing platforms and instruments, sharpened the need for the time-continuous, rather
44 than intermittent, blending of observations and models (*Ghil et al.* [1979]; *Bengtsson et*
45 *al.* [1981]) .

46 To better understand this new point of view, consider a sequence of observations at
47 discrete times $\{t_k : t_0 \leq t_k \leq t_K\}$ of a scalar or vector variable $x(t_k)$ or $\mathbf{x}(t_k)$. The vector
48 variable $\mathbf{x}(t_k)$ will represent the spatially discretized values of a geophysical field, such
49 as temperatures in NWP or PSD values in the radiation belts. *Wiener* [1949] defined
50 filtering, smoothing and prediction of this variable $\mathbf{x}(t_k)$ as its estimate at: (i) the final
51 observing time t_K ; (ii) at all t_k over the observation interval $t_0 \leq t_k \leq t_K$; and (iii) at any
52 time after the final observation, $t_K < t_k$.

53 In the real-time prediction problem, it is pretty easy to convince oneself that — under
54 fairly general hypotheses on the process to be predicted and given observations up to time
55 t_K — the best use one can make of the observations is to estimate as well as one can, with
56 the knowledge one has, the state at the initial prediction time, i.e., at t_K . This is precisely
57 how the so-called forecast-assimilation cycle proceeds in NWP; such an operational NWP
58 cycle is illustrated in Figs 1(a,b).

59 Panel (a) of the figure shows the traditional blending of data and model, at the synoptic
60 times, used from the beginnings of data assimilation in the 1960s until the late 1970s and
61 early 1980s; such data windows are still used in so-called variational methods of data
62 assimilation (e.g., *Courtier and Talagrand* [1987]). Panel (b) outlines the more recent
63 approach, in which data are assimilated at any model time step at which they become
64 available; this sequential approach includes a great variety of methods that generally fall
65 these days in the broad class of Kalman-type filters (e.g., *Jazwinski* [1970] or *Gelb* [1974]).

66 In space physics, given the total absence — at any given time of day or night — of
67 synoptic data that cover the entire domain of interest, it is natural to start relying on the
68 time-continuous approach of data assimilation. This approach can address two types of

69 applications that are of primary interest for the satellite design and operations community:
70 nowcasting and short-term forecasting, as well as long-term reanalysis. For both types of
71 applications, be it in a research or operational mode, data should be assimilated from the
72 operating space platforms and instruments, when and where they become available.

73 The nowcasting applications help address issues linked to the state of the space envi-
74 ronment’s radiation properties at a given time and location, and thus provide post facto
75 insight into the possible causes of particular anomalies. Moreover, a satellite operator
76 could take preventive action, based on a reliable short-term forecast of the space environ-
77 ment, if a satellite is threatened; given the current lack of such reliable forecasts, such
78 action is not a widespread practice at this time.

79 In a research mode, one can also consider the smoothing problem, which produces
80 “movies” of the plasma properties, particle distribution functions, the magnetic and elec-
81 tric fields or the wave environment over the entire lifetime of a satellite or of a group of
82 spacecraft, which may last over several solar cycles. Such a movie can help determine the
83 average state and extreme conditions in a certain part of the space environment, and can
84 be turned into satellite specifications. Our proposed improvement of assimilation method-
85 ology should thus help both operational and research aspects of space physics by providing
86 better estimates of the radiation environment whenever observations are available.

1.2. The need for a log-density formulation

87 A striking feature of the radiation belts is that values of observed electron fluxes and
88 modeled PSD vary by several orders of magnitude, and that the corresponding error
89 distributions are, therefore, not Gaussian. Still, standard data assimilation methods —

90 such as the Kalman filter and its various adaptations to large-dimensional and nonlinear
91 problems — are essentially based on least-squares minimization of Gaussian errors.

92 Even though this mismatch between the nature of the data and that of the method leads
93 to substantial problems when applying standard assimilation methods to the radiation
94 belts, there have been very few investigations to address these issues; we cite here the two
95 that we are aware of: first, *Naehr and Toffoletto* [2005] relied on a log-based transformation
96 of the PSD and on an extended Kalman filter to study sequential filter performance on
97 synthetic data. Next, *O'Brien and Guild* [2010] proposed a variational data assimilation
98 method based on a Maximum Likelihood Ensemble Filter (MLEF, *Zupanski* [2005]) that
99 also uses a log-based transformation for both the measurements and the model state
100 vector.

101 In the spirit of these two studies, we explore alternative ways to make Kalman-filter-type
102 methods more efficient for use in PSD assimilation for the radiation belts by relying on
103 an one-dimensional (1-D) version of the UCLA Versatile Electron Radiation Belt (VERB)
104 diffusion model (*Shprits et al.* [2005], *Subbotin and Shprits* [2008]; see also Sec. 2 herein)
105 and on observations from multiple satellites (Sec. 3). We introduce a log-normal PSD
106 transformation in the UCLA VERB 1-D code to derive an analytical model equation for
107 the transformed variable (Sec. 5) and use it in our extended Kalman filter formulation of
108 Sec. 4.

109 First, we analyze the performance of the log-normal Kalman filter so derived on syn-
110 thetic data in the “fraternal-twin” experiments of Sec. 6.1, and then apply it to spacecraft
111 PSD measurements in Sec. 6.2. We conclude in Sec. 7 by analyzing under which condi-
112 tions this log-normal formulation improves the accuracy of the reanalysis and prediction

113 of the PSD field, as inferred from the variance of the Kalman filter’s innovation sequence.
 114 The main factor influencing the performance of the log-normal filter is the ratio of the
 115 model error to the observational error. When this ratio is sufficiently small but not neg-
 116 ligible, the log-normal formulation produces a more efficient modification of the model
 117 forecast in observation-void regions and better prediction, too.

118

2. The UCLA VERB Code

119 The version of the UCLA VERB code that we use here provide a 1-D description of
 120 the time evolution of the PSD $f = f(L^*, t; \mu, J)$ in the Van Allen radiation belts, at
 121 fixed values of the adiabatic invariants μ and J . The radial variable L^* is the distance
 122 — in the equatorial plane, measured in Earth radii R_E — from the center of the Earth
 123 to the magnetic field line around which the electron moves at time t , assuming that
 124 the instantaneous magnetic field is adjusted adiabatically to a pure-dipole configuration.
 125 In this study, the *Tsyganenko* [1989] T89 magnetic field model has been used to derive
 126 electron fluxes at a particular L^* -value. For simplicity from now on in the text and figures
 127 we drop the superscript and refer to this variable simply as L : both the radiation belt
 128 model and all satellite data are computed in L^* .

The PSD evolution in time is then governed by the following parabolic partial differential equation [*Shultz and Lanzerotti*, 1974; *Walt*, 1994] :

$$\frac{\partial f}{\partial t} = L^2 \frac{\partial}{\partial L} (L^{-2} D_{LL} \frac{\partial f}{\partial L}) - \frac{f}{\tau_L}. \quad (1)$$

129 The radial diffusion term in Eq. (1) represents the violation of the third adiabatic invari-
 130 ant, while the net effect of sources and losses due to violations of the μ and J invariants
 131 is modeled in this equation by the linear decay with a characteristic lifetime τ_L .

The parameters D_{LL} and τ_L in Eq. (1) vary rapidly in space and time, and depend on the background plasma density, as well as on the spectral intensity and spatial distribution of various plasma waves; all of these conditions are extremely difficult to specify accurately from limited point measurements. In this study, we adopt a commonly used empirical relationship due to *Brautigam and Albert* [2000] which is based on the magnetic field measurements at $L = 4$ *Lanzerotti and Morgan* [1973] and $L = 6.6$ *Lanzerotti et al.* [1978], between the radial diffusion coefficient D_{LL} and the geomagnetic activity index Kp :

$$D_{LL}(Kp, L) = 10^{(0.506Kp-9.325)} L^{10}; \quad (2)$$

132 this equation applies throughout the outer radiation belt.

133 For the lifetime parameter τ_L , we consider different parameterizations inside and out-
 134 side the plasmasphere. The latter is a region of the inner magnetosphere that contains
 135 relatively cool and dense plasma at low energies; it is populated by the outflow of iono-
 136 spheric plasma along the magnetic field lines, and consists of closed equipotential surfaces.
 137 The plasmopause that separates it from the regions of open equipotential surfaces lies,
 138 under quiet conditions, within the outer belt, at $L_{PP} = 5 - 6R_E$, where R_E is the Earth's
 139 radius. Under quiet conditions, the outer belt lies at about $3.5 - 6R_E$ and the inner belt at
 140 about $1 - 2.5R_E$, starting just above the ionosphere. Magnetospheric storms deplete the
 141 plasmasphere and L_{PP} can sink to 3 or, for particularly strong storms, even $2R_E$, [*Baker*
 142 *et al.*, 2004].

As described in *Kondrashov et al.* [2007], distinct loss processes operate inside and outside of the plasmasphere, and so we account for them separately in the physical model. Inside we assume that $\tau_{LI} = 10$ days is constant in time, while outside we take

$$\tau_{LO} = \zeta/Kp(t). \quad (3)$$

143 To discretize numerically Eq. (1), we use standard second-order centered difference ap-
 144 proximations for spatial derivatives. We also utilize a fully implicit numerical method to
 145 advance the solution in time, which allows one to use much larger time steps than explicit
 146 schemes do.

3. Spacecraft Observations

147 This study covers a time interval of 120 consecutive days that starts on July 30, 1990,
 148 and includes measurements from four space missions: the Combined Release and Radi-
 149 ation Effects Satellite (CRRES), GEO-1989 (hereafter referred to as GEO), GPS NS18
 150 (hereafter GPS), and Akebono. To perform data assimilation of the PSD distribution de-
 151 rived from the electron flux observations measured by the various spacecraft instruments,
 152 we need first to calculate the PSD in the appropriate phase-space coordinates (μ, K, L) ;
 153 here μ is the first adiabatic invariant, while K is a combination of the first two adiabatic
 154 invariants that is independent of the particle mass and charge.

155 The Kp data set is taken from the World Data Center for Geomagnetism in Kyoto,
 156 Japan, <http://swdcdb.kugi.kyoto-u.ac.jp/aedir/>. The T89 model is specified by the Kp -
 157 value and is valid only for relatively modest activity levels. Recently, *Ni et al.* [2009]
 158 have compared and mutually calibrated PSD data from the CRRES Medium Electron
 159 A (MEA) observations and those from the polar-orbiting Akebono Radiation Monitor

160 (RDM) by using the T89 model; they found in general good agreement between the PSD
 161 values inferred from the two sets of observations.

162 Recent, improved models of the magnetic field include parameterizations that use also
 163 *Dst* and solar-wind measurements. Though the latter are not generally available for the
 164 CRRES time period, *Kondrashov et al.* [2010] showed that Singular Spectrum Analysis
 165 can be used to fill in large gaps in past solar-wind and IMF data.

4. The Extended Kalman Filter (EKF)

166 In this section, we review the Kalman filter as applied to data assimilation in the
 167 radiation belts, following *Kondrashov et al.* [2007], *Shprits et al.* [2007], *Ni et al.* [2009],
 168 *Koller et al.* [2007] and *Daae et al.* [2011]. The summary here uses the filter’s presentation
 169 for partial differential equations in the geosciences, as introduced by *Ghil et al.* [1981]
 170 and reviewed by *Ghil and Malanotte-Rizzoli* [1991]; in this presentation, both time and
 171 space have been discretized by finite differences.

The time evolution of the state vector $\mathbf{x}_k^{\text{f,t}} = x^{\text{f,t}}(k, \Delta t)$ is assumed to be governed by the numerically discretized system of equations whose right-hand side (RHS) is denoted by $\mathbf{F} = \mathbf{F}(\mathbf{x})$; here superscript “t” refers to *true*, “f” refers to model *forecast*, and k is a discretized time index. If the system is nonlinear, $\mathbf{F} = \mathbf{F}(\mathbf{x})$ has to be linearized to yield the model matrix \mathbf{M} that will be used in advancing the forecast error covariances. Furthermore, the true state differs from the model forecast by a random error ϵ^{m} :

$$\mathbf{x}_k^{\text{f}} = \mathbf{F}_k(\mathbf{x}_{k-1}^{\text{f}}), \quad (4)$$

$$\mathbf{x}_k^{\text{t}} = \mathbf{F}_k(\mathbf{x}_{k-1}^{\text{t}}) + \epsilon_k^{\text{m}}, \quad (5)$$

$$\mathbf{M} = \frac{\partial \mathbf{F}}{\partial \mathbf{x}}. \quad (6)$$

172 For the radiation belt model of Eq. (1), the components of the state vector \mathbf{x}_k are the
 173 PSD values at the discretized grid points in the independent variable L ; since the partial
 174 differential operator in (1) is linear, no linearization, as in Eq. (5), seems to be required.
 175 In the next section, though, we will encounter a nonlinear version of this equation, thus
 176 justifying the use of the full set of Eqs. (4)–(6).

177 The model noise ϵ^m accounts for the net errors due to inaccurate model physics,
 178 such as errors in forcing, boundary conditions, numerical discretization, and subgrid-
 179 scale processes. Commonly, the column vector ϵ^m is assumed to be a Gaussian white-
 180 noise sequence, with mean zero and model-error covariance matrix \mathbf{Q} , $E[\epsilon_k^m] = 0$ and
 181 $E[\epsilon_k^m \epsilon_l^{mT}] = \mathbf{Q}_k \delta_{kl}$, where E is the expectation operator, superscript “T” denotes the
 182 transpose, and δ_{kl} is the Kronecker delta.

The observations \mathbf{y}_k^o of the true system, where superscript “o” refers to “observed,” are
 also perturbed by Gaussian white noise ϵ_k^o with mean zero and given covariance matrix
 \mathbf{R} , $E[\epsilon_k^o \epsilon_l^{oT}] = \mathbf{R}_k \delta_{kl}$:

$$\mathbf{y}_k^o = \mathbf{H}_k \mathbf{x}_k^t + \epsilon_k^o. \quad (7)$$

183 The observation matrix \mathbf{H}_k accounts for the fact that usually the dimension of \mathbf{y}_k^o is less
 184 than the dimension of \mathbf{x}_k^t , i.e., at any given time observations are not available for all grid
 185 points. It is often also the case that the values of the PSD or other state variable are not
 186 directly observable, and it is only some linear or nonlinear combination of such variables,
 187 such as weighted integrals over phase space, that can be measured.

The Kalman filter and its variants are sequential data assimilation methods. For given
 model and observation error covariances, \mathbf{Q} and \mathbf{R} , the filter combines the model forecast
 with the observations so as to obtain the *analysis* that is closest in a least-square sense

to the truth. The gain matrix \mathbf{K}_k in Eq. (8) represents the *optimal* weights given to the observations in updating the model forecast, based on this least-square minimization:

$$\mathbf{x}_k^a = \mathbf{x}_k^f + \mathbf{K}(y_k^o - \mathbf{H}\mathbf{x}_k^f), \quad (8)$$

$$\mathbf{K} = \mathbf{P}^f \mathbf{H}^T (\mathbf{H} \mathbf{P}^f \mathbf{H}^T + \mathbf{R})^{-1}, \quad (9)$$

$$\mathbf{P}_k^f = \mathbf{M}_k \mathbf{P}_{k-1}^f \mathbf{M}_k^T + \mathbf{Q}, \quad (10)$$

$$\mathbf{P}_k^a = (\mathbf{I} - \mathbf{K}\mathbf{H})\mathbf{P}_k^f. \quad (11)$$

188 The error-covariance matrices $\mathbf{P}^{f,a}$ are the time-dependent error estimates for the fore-
 189 cast and the analysis, respectively. One expects, from Eqs. (10) and (11), that the analysis
 190 error be smaller than the forecast error, cf. *Ghil et al.* [1981] and *Carrassi et al.* [2008].
 191 The sequential estimator for nonlinear systems that uses the linear dynamics operator
 192 \mathbf{M}_k in the quadratic equation (10) for advancing the model covariances \mathbf{P}_k^f in time, while
 193 preserving the nonlinear evolution (4) of the state itself, is called the *extended Kalman*
 194 *filter* (EKF: *Jazwinski* [1970], *Gelb* [1974], *Miller et al.* [1994]). To estimate poorly known
 195 parameters of the system, Kalman filter can be applied to state vector augmented with
 196 the parameters values (*Kondrashov et al.* [2007,2008]).

197 It is logical to assume that the PSD is log-normally distributed since it is always posi-
 198 tive, and generally its variations — as measured, for instance, by the standard deviation
 199 — increase as its mean value increases. Normally distributed variables, on the other hand,
 200 can be negative and have a standard deviation that does not change as the mean changes.
 201 Log-normal errors arise when variation sources accumulate multiplicatively, whereas nor-
 202 mal errors arise when these sources are additive [*Crow and Shimizu*, 1988].

203 By assuming a log-normal distribution of errors at each location and errors that are
 204 uncorrelated between different locations, both \mathbf{Q} and \mathbf{R} can be specified as diagonal ma-
 205 trices, and their diagonal terms can be taken simply as $\alpha_{o,m} f_{o,m}^2$, where $f_{o,m}^2$ is the observed
 206 or modeled PSD value, and $\alpha_{o,m}$ is a specified factor that corresponds to observational or
 207 model error. Note that the exact values of $\alpha_{o,m}$ are not important: it is their respective
 208 ratio that determines the weights given to the observations vs. the model solution in the
 209 analysis, or update, step of the data assimilation. In this study, we follow approach of *Ni*
 210 *et al.* [2009] with the value of α_o depending on the intercalibration of satellite data; we
 211 use $\alpha_o = 200$ for Akebono and $\alpha_o = 400$ for GEO, while $\alpha_m = 25$.

212 Due to their reliance on least-squares minimization, the EKF and other Kalman-type
 213 filters may not be efficient in modifying the model forecast at locations where the obser-
 214 vations and forecast differ by several orders of magnitude. In the next section (Sec. 5), we
 215 use a log-normal transformation of variables to derive the model equation for $\log(f)$ and
 216 the corresponding EKF. We then study in Sec. 6.1 the performance of the log-normal EKF
 217 for synthetic data in “fraternal-twin” assimilation experiments — when the true evolution
 218 of the system is known — and next, in Sec. 6.2, for actual space-borne observational data.

219

5. Log-Normal Model and Filter

By introducing the new variable $S = \log(f)$ in Eq. (1) and using the chain rule for partial derivatives in time and space, one can easily obtain the following evolution equation for the log-transformed PSD:

$$\frac{\partial S}{\partial t} = L^2 \frac{\partial}{\partial L} \left(\frac{1}{L^2} D_{LL} \frac{\partial S}{\partial L} \right) - \frac{1}{\tau_L} + D_{LL} \left(\frac{\partial S}{\partial L} \right)^2. \quad (12)$$

220 The first two terms on the RHS of the log-transformed Eq. (12) correspond to radial
 221 diffusion and losses respectively, as in Eq. (1). Even though the original Eq. (1) is linear,
 222 the last term on the RHS of Eq. (12), $(\partial S/\partial L)^2$, is due to the nonlinear transformation of
 223 variables and requires special attention in the numerical solution. This term becomes im-
 224 portant in locations where strong PSD gradients occur; it can be understood as nonlinear
 225 advection of S with a velocity that depends on its gradient.

226 Note that our methodology is distinctly different from that of *Naehr and Toffoletto*
 227 [2005], where a log-normal transformation is applied to the *numerically discretized* equa-
 228 tion for f . We derive instead the analytical equation (12) for the evolution of $S = \log(f)$;
 229 this equation does not depend on the details of a particular numerical scheme for solving
 230 the f -equation (1).

231 Note that the log-transformed Eq. (12) and the original equation (1) should both yield
 232 the same solution $f(t, L)$ — to within the accuracy of the spatial discretization and time
 233 integration scheme — when solved numerically. The numerical solution of the original
 234 Eq. (1) is typically a smooth monotone function in space, but “naive” approximation in
 235 space of the quadratic term in Eq. (12) — for example by using centered differences — will
 236 result in spurious local extrema when integrated numerically. This Gibbs phenomenon
 237 ultimately leads to unstable solutions and — in order to avoid such numerical instabilities
 238 and preserve the monotonicity of the solutions of the log-transformed Eq. (12) — we use
 239 an upwind approximation of $(\partial S/\partial L)^2$ that is second-order in space and total-variation
 240 diminishing (TVD: *Harten* [1983]). The TVD scheme, when combined with the implicit
 241 time integration, guarantees stable numerical solutions of Eq. (12). To solve numerically

242 either Eq. (1) or Eq. (12), we use a uniform grid of 100 points in L ; the number of grid
 243 points also determines the dimension of the state vector in the Kalman filter formulation.

244 Steep gradients are key features of the radiation belts, and there is therefore an addi-
 245 tional improvement in numerical performance obtained by recasting the diffusion problem
 246 of Eq. (1) in terms of the transformed variable $S = \log(\text{PSD})$: doing so helps avoid non-
 247 physical negative values, which may arise in numerical schemes that solve Eq. (1) in the
 248 original variable $f = \text{PSD}$. There is of course a trade-off in difficulty, since the log-normal
 249 model requires one to solve the nonlinear Eq. (12); this will also present a special chal-
 250 lenge when applying the proposed methodology to the 3-D VERB code (*Subbotin and*
 251 *Shprits, 2008*) that describes diffusion in energy, pitch angle and L .

252 Since Eq. (12) is nonlinear in S , we also need to linearize the model in order to imple-
 253 ment the EKF of Eqs. (4)–(11). Moreover, for the EKF implementation, observational
 254 and model errors for f are modified in a manner appropriate for log-transformed vari-
 255 ables by setting the diagonal elements \mathbf{Q} and \mathbf{R} equal to $\log(1 + \alpha_m)$ and to $\log(1 + \alpha_o)$,
 256 respectively [*Crow and Shimizu, 1988*].

6. Numerical Results

6.1. “Fraternal-twin” experiments

257 To compare the log-normal EKF scheme of Sec. 5 with the standard EKF implemen-
 258 tation of Sec. 4, we conduct so-called “fraternal-twin” experiments in which both the
 259 “true” solution, from which the observations are drawn, and the forecast are produced by
 260 the same model, but with different values of the lifetime parameters in Eqs. (1) and (12).
 261 This type of experiment is a harder test for a given assimilation method than a so-called

262 “identical-twin” experiment, in which the model used for the assimilation of partial data
263 is identical to the one used to generate the data, and only the initial state may differ.

264 We obtain our true PSD distribution from a model run with $\tau_{LI} = 10$ days and $\zeta = 5$
265 days, cf. Eq. (3) and Fig. 2a; this run is also used to generate synthetic observations
266 along the tracks of the GPS and GEO satellites with a 10-min resolution, as plotted in
267 Fig. 2c. Our goal is to recover the true solution by assimilating these observations into a
268 model simulation with an “incorrect” set of parameters, equal to $\tau_{LI} = 10$ days and $\zeta = 1$
269 day; these values correspond to higher losses, as shown in Fig. 2b.

270 The results of assimilating the synthetic data from Fig. 2c by applying the standard
271 EKF formulation are plotted in Fig. 3a. The plot shows that even though assimilating
272 this data set drives the model forecast with the wrong parameter values towards the true
273 model’s solution, significant differences remain, cf. Fig. 3c. These differences are largest
274 for $4 \leq L \leq 6$, i.e. in the heart of the outer radiation belt, where the PSD gradients are
275 strongest in Fig. 2b.

276 When using the log-normal EKF of Sec. 5, on the other hand, our data assimilation
277 reduces the model forecast error much more efficiently in the region of strong PSD gra-
278 dients. This can be clearly seen by comparing Figs. 3b and 3d with the preceding Figs.
279 3a and 3c; it is also confirmed by the time-averaged analysis error in PSD values, for the
280 standard EKF formulation and the log-normal one, as shown in Fig. 4b.

281 Since both the original model equation (1) and the log-transformed equation (12) yield
282 identical solutions for the PSD field in the absence of data assimilation, the difference in
283 the results can only be due to the change in the data assimilation scheme, as outlined and
284 explained in Sec. 5.

In addition to comparing the analysis errors, as discussed above, another useful and readily available means for assessing an assimilation scheme is studying the *innovation sequence*:

$$\mathbf{z}_k \equiv \mathbf{y}_k^o - \mathbf{H}\mathbf{x}_k^f, \quad (13)$$

285 which appears in the updating equation (8). The importance of considering the inno-
 286 vations \mathbf{z}_k in sequential estimation — i.e., the time series of the differences between the
 287 observations and the model forecast — was noted already by *Kailath* [1968] and was
 288 emphasized recently by *Fukumori* [2006] in the geophysical context.

289 The innovation vector represents the filter’s correction to the model dynamics. For a
 290 linear system with known coefficients and known noise covariances, the *innovation property*
 291 of the Kalman filter states that the innovation sequence has zero mean and is white in
 292 time, i.e., $E[\mathbf{z}_k^T \mathbf{z}_l] = 0$ for $k \neq l$; this means simply that the filter extracts, at each time
 293 step, any and all useful information from the observations. *Dee et al.* [1985], for instance,
 294 have used systematically deviations from this property to infer unknown error covariances
 295 \mathbf{Q} and \mathbf{R} in the shallow-water model of *Ghil et al.* [1981]. In the space-plasma context,
 296 *Koller et al.* [2007] and *Shprits et al.* [2007] showed how the nonzero time mean of the
 297 innovation sequence can point to missing physics in the model’s competing processes of
 298 losses and local acceleration.

299 Here we show how inspection of the innovation sequence can be used to diagnose the
 300 performance of a data assimilation scheme, even in the case of a nonlinear problem,
 301 like Eq. (12). As the forecast \mathbf{x}_k^f does not yet utilize the upcoming observations \mathbf{y}_k^o ,
 302 the variance of the sequence \mathbf{z}_k provides an objective measure of *prediction skill* with
 303 respect to independent observations: a perfect model would predict exactly the incoming

304 observation vector, so that $\mathbf{y}_k^o = \mathbf{H}\mathbf{x}_k^f$, while a particularly poor model might differ from
 305 the observations more than their long-term mean, i.e., than the climatology of the system.

306 The prediction skill is thus defined here as the variance $E[\mathbf{z}_k^T \mathbf{z}_k]$ and it is plotted in
 307 Fig. 4a for both the standard EKF formulation (red curve) and the log-normal one (blue
 308 curve); for the latter Eq. (13) has been converted into PSD space to make the two
 309 estimates comparable. In addition, the straight model simulation of Fig. 2b (dashed
 310 black curve), without data assimilation, has been plotted as well. The prediction skill
 311 of these three types of forecast is compared in turn to the total variance of the PSD
 312 observations obtained from the control run of Fig. 2a; the latter should be reduced by
 313 the forecast model's interpolating the sparse data, even though this model is not perfect.

314 In practice, we see that for our fraternal-twin experiments, the model simulation with
 315 the wrong parameter value of $\zeta = 1$ day does not yield any useful prediction skill, as the
 316 variance of its innovation sequence is even higher than the variance of the “observations,”
 317 i.e. of the control run. On the other hand, both EKF formulations reduce the variance
 318 of the innovation sequence, while the log-normal formulation has a substantially better
 319 prediction skill than the standard EKF at all L -shells, cf. Fig. 4a. In addition, the
 320 analysis obtained by the log-normal formulation has a substantially lower time-averaged
 321 error with a truth (cf. Fig. 3c,d) at all L -shells, cf. Fig. 4b.

322 The results in Figs. 2–4 have been obtained with observational errors set much larger
 323 than the model error: $\alpha_m = 25$ and $\alpha_o = 100\alpha_m$. In this situation, the EKF can more
 324 easily correct the model forecast's state-vector components at grid points away from the
 325 observation sites [*Ghil and Malanotte-Rizzoli, 1991*]: When the model is assumed to be
 326 more accurate than the observations, then the EKF's weights in Eq. (9) for such locations

327 are non-negligible, due to the spatial correlations inferred from the error covariance matrix.
328 The log-normal formulation, due to its capability to capture better very large variations in
329 PSD values, allows for much larger corrections of the model forecast where the gradients
330 are steepest.

331 When the observational and model errors are comparable, say $\alpha_m = \alpha_o$, both formula-
332 tions yield very similar data assimilation results, with but small differences in prediction
333 skill: the smallness of the differences apparent in Fig. 5 is largely due to the fact that
334 the model forecast is modified to a much lesser extent at grid points away from the ob-
335 servation sites. Finally, when the model error is much larger than the observational error,
336 the EKF approximates the “direct-insertion method,” in which the observations simply
337 replace the model forecast at all the points where observations are taken. In this case
338 (not shown), the EKF results are the same, regardless of the formulation chosen. At the
339 opposite end of the error ratio scale, when the model errors are negligible, the EKF will
340 ignore the observations completely.

341 These results suggest that there is a certain range of ratios between observational and
342 model errors within which the log-normal EKF formulation will perform better than
343 the standard one. In particular, based on our fraternal-twin experiments with synthetic
344 data, the log-normal formulation of the EKF is expected to perform better when the
345 observational errors are larger than the model errors. In the next section, we verify these
346 results by assimilating actual satellite data, whose errors are quite large.

6.2. Spacecraft data assimilation

347 In this section, we compare the standard and the log-normal EKF formulations by assim-
348 ilating PSD data derived from measurements on-board the Akebono and GEO spacecraft.

349 These are assimilated into the VERB-1D code with the loss parameters $\tau_{LI} = 10$ days
 350 and $\zeta = 5$ days; see Eq. (3) and Fig. 2a. Unlike in the fraternal-twin experiment of
 351 the preceding section, here we do not know the continuous spatio-temporal evolution of
 352 the true PSD field. Instead, we will consider as a comparison benchmark independent,
 353 high-quality observations from the CRRES spacecraft, with more complete coverage in
 354 time and space, as shown in Fig. 6a.

355 Based on intercalibration of PSD data we assume $\alpha_m = 25$ for the model error in the
 356 VERB-1D code, while we take observational error $\alpha_o = 200$ for Akebono and $\alpha_o = 400$ for
 357 GEO. Such a choice of error parameters allows the Kalman filter to modify efficiently the
 358 full state vector, as described in Sec. 6.1. In agreement with our synthetic-data results
 359 there, the assimilation results (not shown) using the PSD data from the CRRES mission
 360 (shown in Fig. 6a) and from the GPS satellite (not shown) do not depend on the EKF
 361 formulation, since these two data sets are of higher quality and have smaller observational
 362 errors than the GEO and Akebono data.

363 First, we assimilate the AKEBONO RDM measurements which do not include the near-
 364 Earth region of steep PSD gradients, $L \leq 3$; the RDM observations are plotted in Fig.
 365 6b. The assimilation results for the standard EKF formulation (Fig. 6c) have several non-
 366 physical PSD maxima at $L \approx 3$; these maxima are absent from the CRRES observations
 367 in Fig. 6a. The results for the log-normal EKF formulation in Fig. 6d, on the other hand,
 368 yield a smooth PSD field in much better agreement with the CRRES data of Fig. 6a.

369 The prediction skill is shown in Fig. 7 and it is improved by both EKF filter formula-
 370 tions in comparison with the model simulation without the benefit of data assimilation.

371 However, the skill for the log-normal EKF is only modestly better at all L -values, i.e. the
372 innovation variance is somewhat smaller than for the standard EKF.

373 Unlike in the Akebono case, the GEO measurements cover only a very narrow L -range,
374 at $L \approx 5$; see Fig. 8b. Such a limited data set presents a greater challenge for the EKF in
375 realistically reconstructing the PSD profile at low L -shells, far away from the observations
376 points, cf. 7c. Even in this case, the prediction skill of the log-normal EKF is uniformly
377 better over the L -range sampled by GEO, as can be seen in Fig. 9.

7. Conclusions

378 This study was motivated by the recognition that both simulated and observed phase-
379 space density (PSD) values in the radiation belts are subject to very large spatio-temporal
380 variations, and that variations over several orders of magnitude may not be adequate
381 for standard data assimilation methods based on least-squares minimization of normally
382 distributed errors. We formulated therefore in Sec. 5 a model and filter version using the
383 logarithm of the PSD as the dependent variable.

384 Our “fraternal-twin” experiments in Sec. 6.1 showed that the proposed log-normal
385 formulation of the extended Kalman filter (EKF) can substantially reduce the assimilation
386 errors in regions of steep PSD gradients; see Figs. 2 and 3. The proposed methodology
387 demonstrates the most substantial improvements when model errors are smaller than the
388 observational errors. Such an error ratio allows the log-normal EKF implementation to
389 modify the model forecast very efficiently in observation-void regions; these modifications
390 lead to much better PSD predictions, as inferred from the variance reduction of the
391 innovation sequence in Figs. 4 and 5.

392 These findings have been confirmed by assimilating PSD measurements from the GEO
393 and Akebono satellites (Figs. 6–9), which have large observational errors derived from in-
394 tercalibration studies. In particular, the log-normal EKF applied to Akebono observations
395 yields an assimilated PSD field in which non-physical maxima are absent, according to
396 independent CRRES validation data. In addition, the prediction skill of the log-normal
397 formulation is better for both the GEO and Akebono data. The results of this study
398 should thus be useful to researchers, as well as to spacecraft designers and engineers, in
399 the transition to operational prediction of the near-Earth space environment of satellites
400 and other high-technology systems.

401 Our proposed rescaling methodology holds even greater promise for the data assimi-
402 lation of multiple-satellite measurements for sophisticated, three-dimensional radiation-belt
403 models. Such models describe much better competing loss and source mechanisms than
404 the 1-D VERB code used in this study, thus reducing further model errors.

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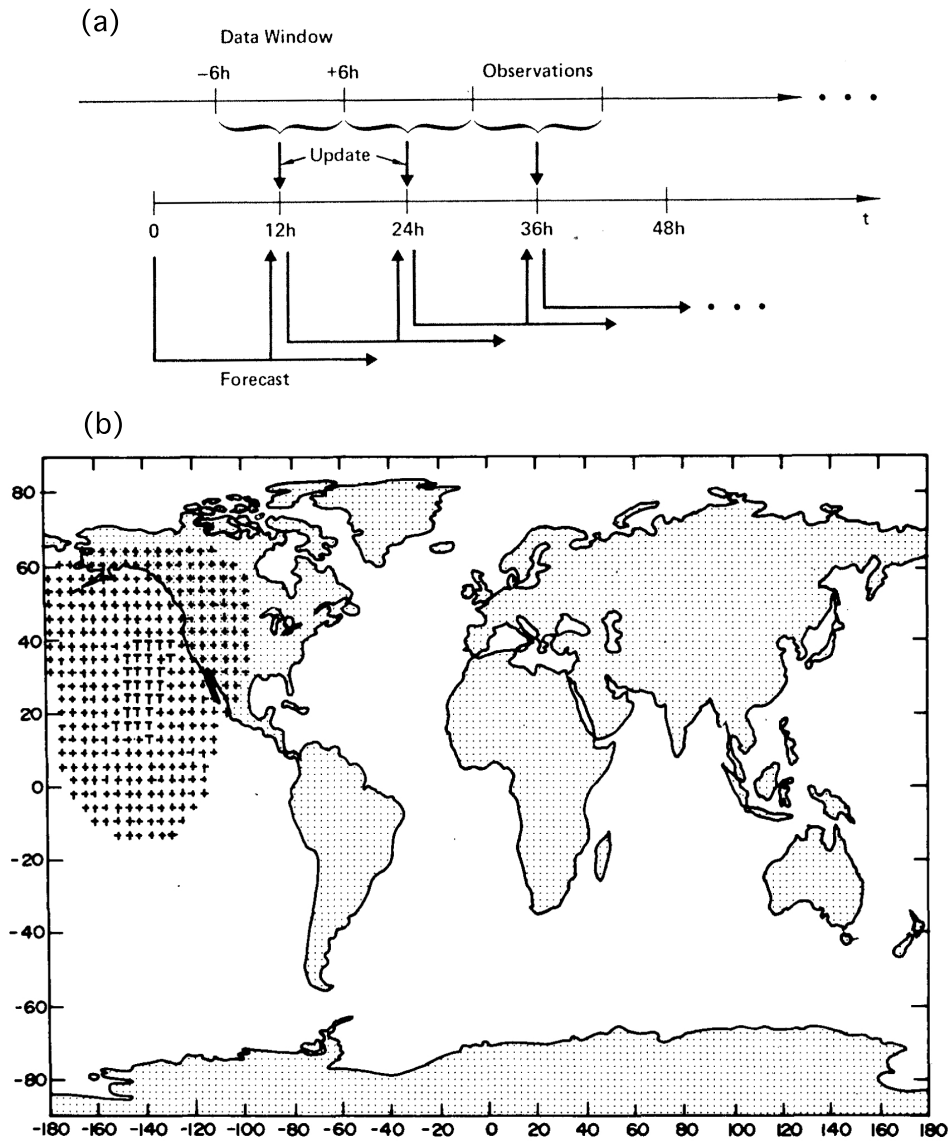


Figure 1. Operational forecast-and-assimilation cycle of a typical weather service that combines the prediction and data assimilation processes. (a) Data are gathered from a “window” of near-past and near-future data, at the synoptic times, 12 h apart; (b) data are assimilated as they become available, at any model time step. In panel (b), the letters ‘T’ stand for the locations at which temperature profiles become available from an infrared satellite sounder at a particular model time step, while the ‘+’ signs indicate grid points that will be affected by those soundings due to the particular sequential filter applied to the soundings. Adapted from (a) *Ghil* [1989], and (b) *Ghil et al.* [1979], respectively.

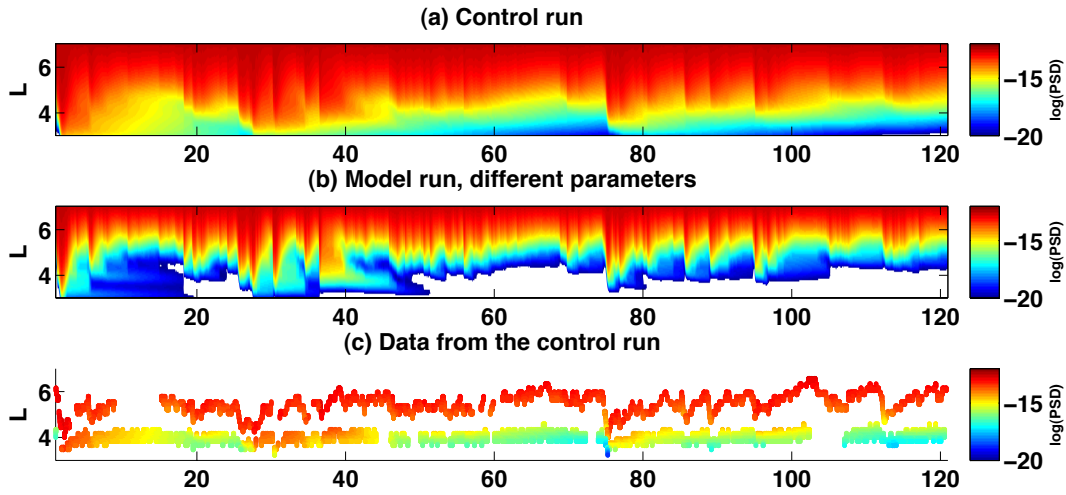


Figure 2. “Fraternal-twin” experiment using synthetic observations from a model simulation with different parameter values. The radiation belt model employs a Kp -dependent lifetime parameterization outside the plasmasphere, with $\tau_{LO} = \zeta/Kp(t)$, cf. Eq. (3). (a) “Truth” given by the model solution with $\zeta = 5$ days, also called the *control run* or *nature run*; (b) model simulation assuming higher losses, with $\zeta = 1$ day; (c) synthetic observations taken from the control run in panel (a).

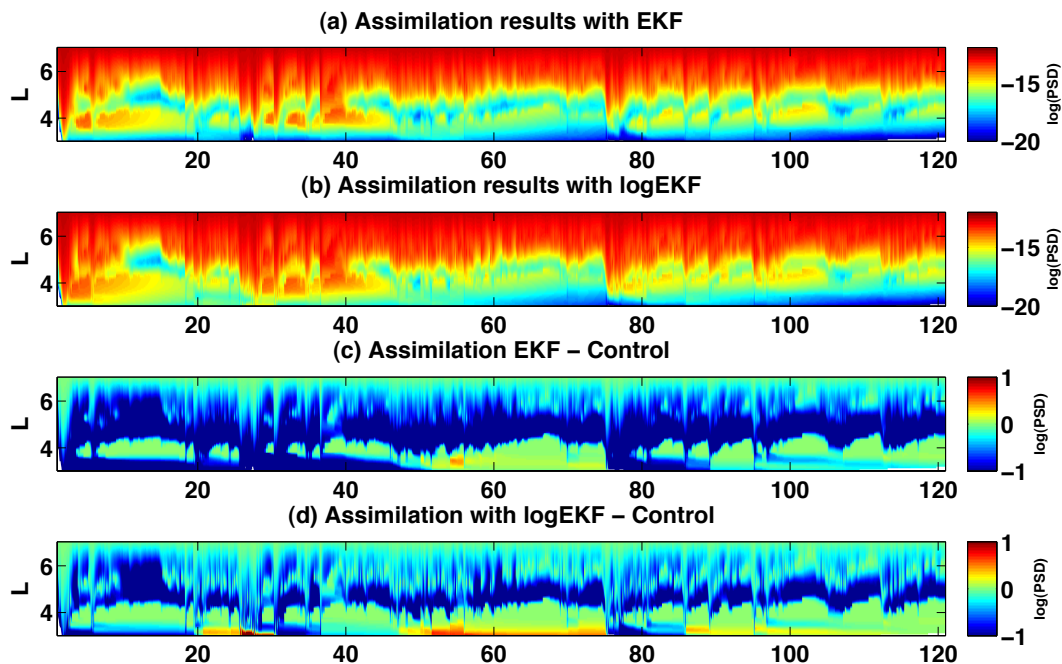


Figure 3. Assimilation results for “fraternal-twin” experiment with forecasts from the model in Fig. 2b and data from the control run in Fig. 2c: (a) using the standard EKF formulation of Sec. 4; (b) same as in panel (a) but for the log-normal EKF of Sec. 5; (c) difference between the assimilation results in panel (a) and the control run of Fig. 2a; (d) difference between the assimilation results in panel (b) and the same control run.

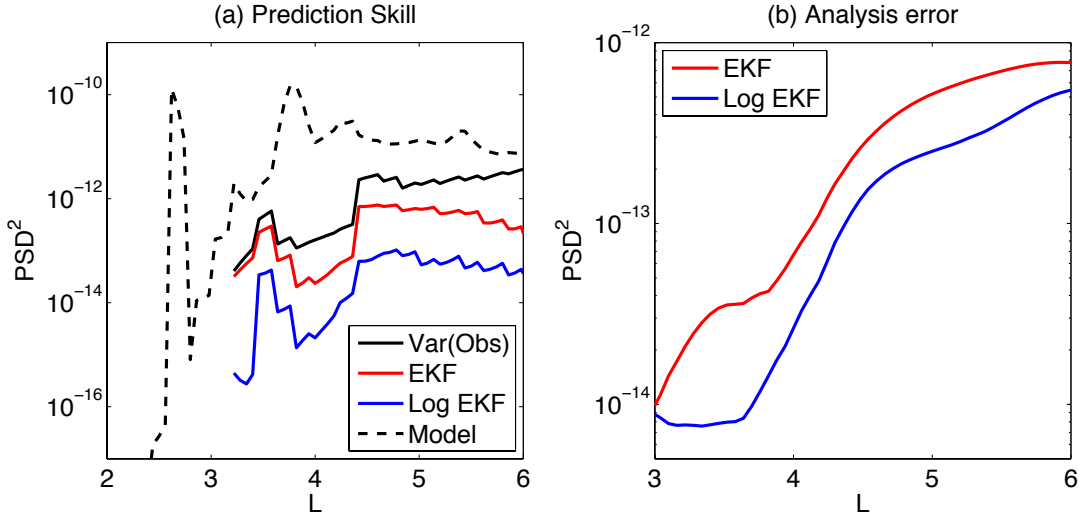


Figure 4. (a) Prediction skill of the models and the sequential estimation methods, defined as variance $E[\mathbf{z}_k^T \mathbf{z}_k]$ of the innovation sequence with \mathbf{z}_k given by Eq. (13); see text for details. Black solid curve: variance of the synthetic PSD observations sampled from the control run, cf. Fig. 2c; black dashed: model simulation with incorrect parameter values and no data, cf. Fig. 2b; red: standard EKF; blue: same for the log-normal EKF but converted into PSD values. Data assimilation clearly improves the models' forecasting ability of the data for all L -values, with the smallest variance of the innovation sequence for the log-normal formulation. These results are for observation errors much larger than the model errors; see text for details. (b) Error computed as time mean of the squared difference between assimilation results and control, given in both cases in terms of PSD values: for the standard EKF (red curve, cf. Fig. 3c), and log-normal EKF (blue curve, cf. Fig. 3d.)

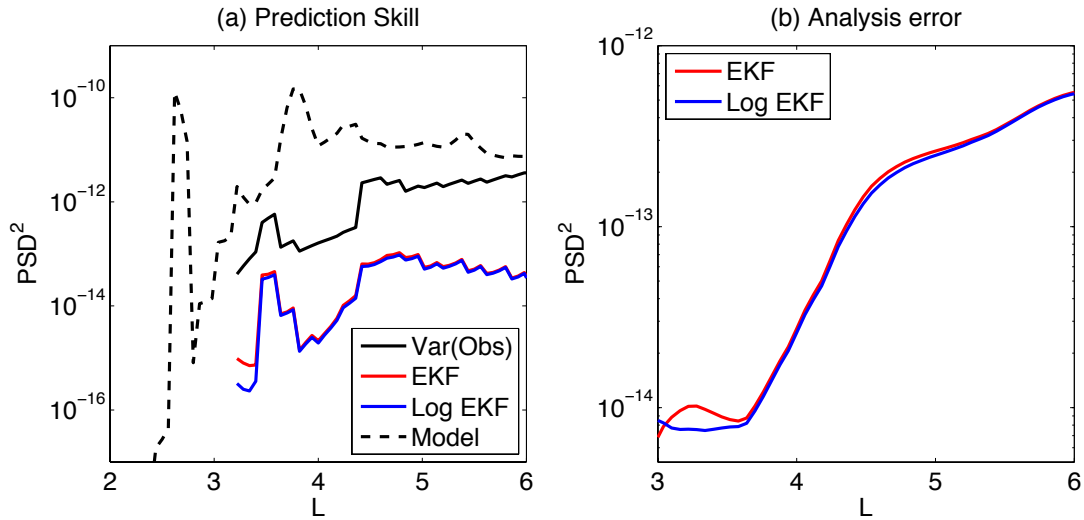


Figure 5. Same as in Fig. 4 but assuming equal model and observational errors; see text for details. In this case, the performance of the standard EKF and the log-normal EKF are quite comparable.

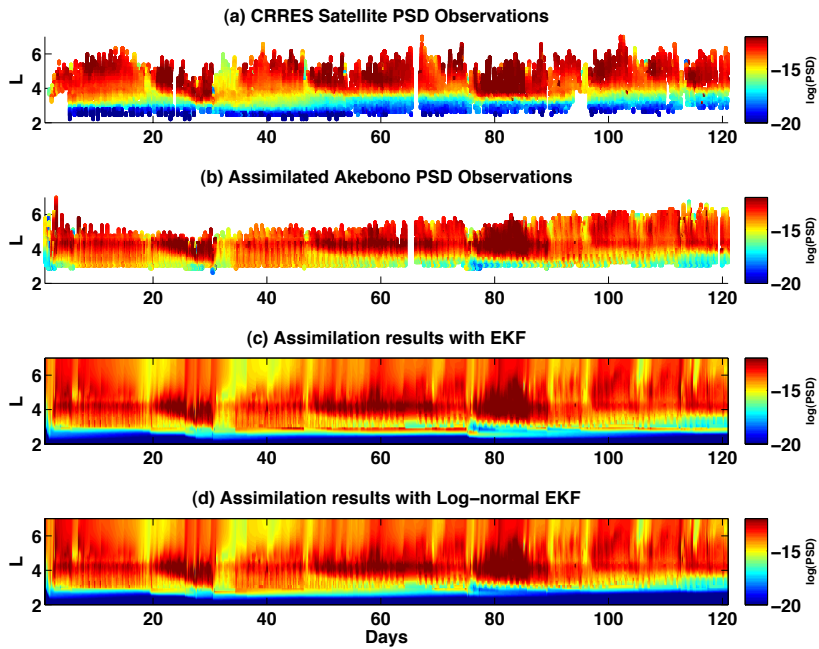


Figure 6. Data assimilation results using real spacecraft data. (a) CRRES PSD observations; (b) assimilated Akebono observations; (c) assimilation results with the standard EKF; (d) assimilation results with the log-normal EKF. The log-normal formulation provides better agreement of the assimilation results with the CRRES observations in the inner belt, $L < 3$, where the PSD gradients are strong.

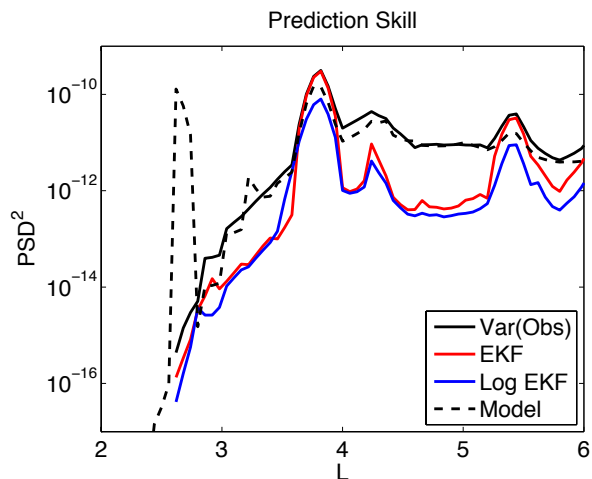


Figure 7. Prediction skill for real spacecraft data. Same color conventions as in Figs. 4a and 5a — solid black: variance of the Akebono PSD observations shown in Fig. 6b; dashed black: model simulation with no data (see again Fig. 2a); red and blue: standard and log-normal EKF, respectively. The Akebono data improve the model’s forecasting ability over all L -values, and the log-normal formulation exhibits the smallest variance of the innovation sequence.

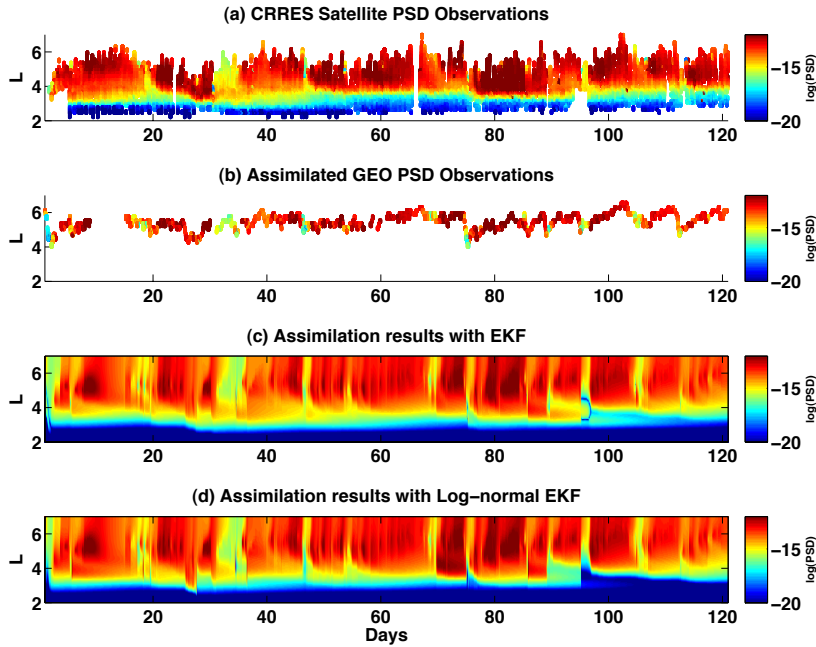


Figure 8. Same as in Fig. 6 but for assimilating observations from the GEO satellite.

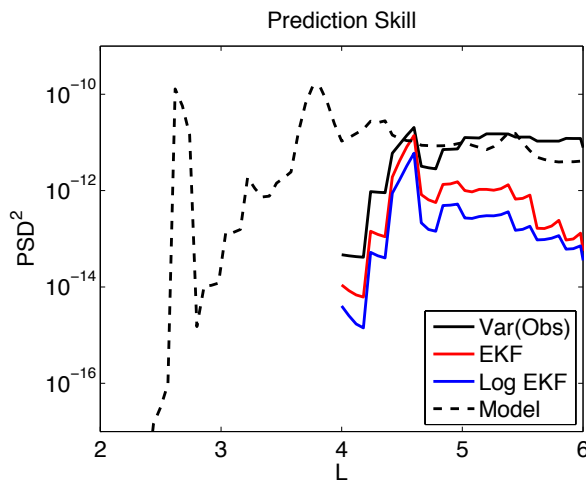


Figure 9. Same as in Fig. 7 but for assimilating observations from the GEO satellite.

The log-normal formulation provides again a smaller variance of the innovation sequence than the standard algorithm, and hence better prediction; compare the blue and red curves, respectively.